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Jeff Mills, "Hydrocarbon Fuels Optimization"

Poster Session - HEDM Conference

(Public Release)

# Hydrocarbon Fuels Optimization

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## Introduction—

Hydrocarbon fuels, as a class, are investigated using the minimal set of parameters determinative of oxidizer-optimized specific impulse and common density-specific-impulse analogues. In this context specific families of promising candidate fuels can be systematically compared, the performance trade-offs among the relevant chemical and physical properties can be quantified, and the general characteristics of mission-tailored fuels can begin to be elucidated.

# Initial Questions—

- Why do I sometimes get relatively poor theoretical performance from a high-energy hydrocarbon?
- Why is RP-1 (kerosene) such a good rocket fuel?
- What are the minimal set of fuel parameters that determine hydrocarbon  $I_{sp}$ ?
- What are the performance trade-offs among these parameters?
- How can I begin to use this information to search for the “optimum” hydrocarbon fuel and reduce “hunt and peck” methods?

# General Concepts / Initial Approximations—

Characteristics of “Good” Rocket Fuels:

- High Heat of Formation
- High “Light-Atom” Content

Specific Impulse:

(Momentum Transfer from Combusted Ejecta)

For an optimal nozzle:

$$I_{sp} \equiv \frac{1}{g} \sqrt{2(H_{chamb.} - H_{exh.})}$$

Sometimes approximated:

$$(I_{sp})_{\text{opt}} \propto \frac{1}{g} \sqrt{2 \Delta H_{\text{comb.}}}$$

or,

$$(I_{sp})_{\text{opt}} \propto \frac{1}{g} \sqrt{2 \Delta H_f}$$

$\Delta H_{\text{comb}}$ —stoichiometric combustion enthalpy  
(per unit mass of exhaust products)  
 $\Delta H_f$ —specific enthalpy of formation

# More Realistic Model Rocket—

One-Dimensional, Adiabatic, Equilibrated, Isentropic Expansion from the Combustion Chamber

Reference Rocket Conditions for  $I_{sp}$  Optimized vs. LOX:

Sea level expansion:

$$P_{\text{chamber}} = 1000 \text{ psi}, P_{\text{exhaust}} = 14.7 \text{ psi}$$

“Vacuum” expansion:

$$P_{\text{chamber}} = 1000 \text{ psi}, \epsilon = 40$$

Sole determinative parameters of LOX-optimized specific impulse and its density variants:

$$(I_{sp})_{\text{opt}} \Leftarrow \Delta H_f(/g) \text{ and } r(\frac{H}{C})$$

$$(\rho_{\text{of}}^a I_{sp})_{\text{opt}} \Leftarrow \Delta H_f(/g), r(\frac{H}{C}), \rho_f, \text{ and } a$$

$\Delta H_f(/g)$ —specific enthalpy of formation of the fuel

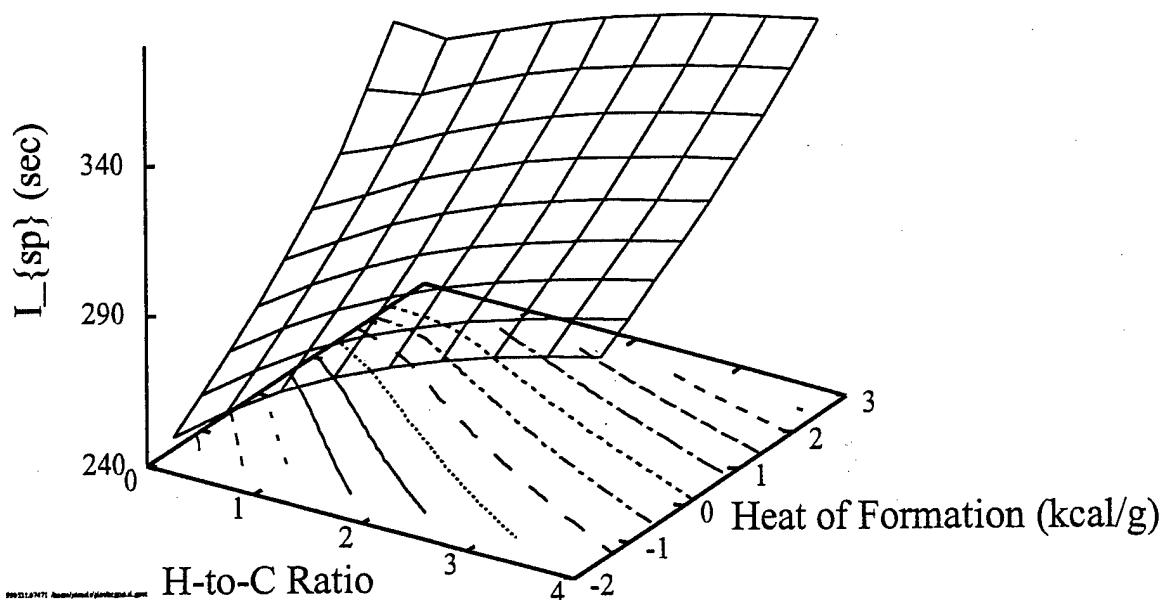
$r(\frac{H}{C})$ —atom ratio of fuel

$\rho_{\text{of}}$ —oxidizer-fuel bulk density

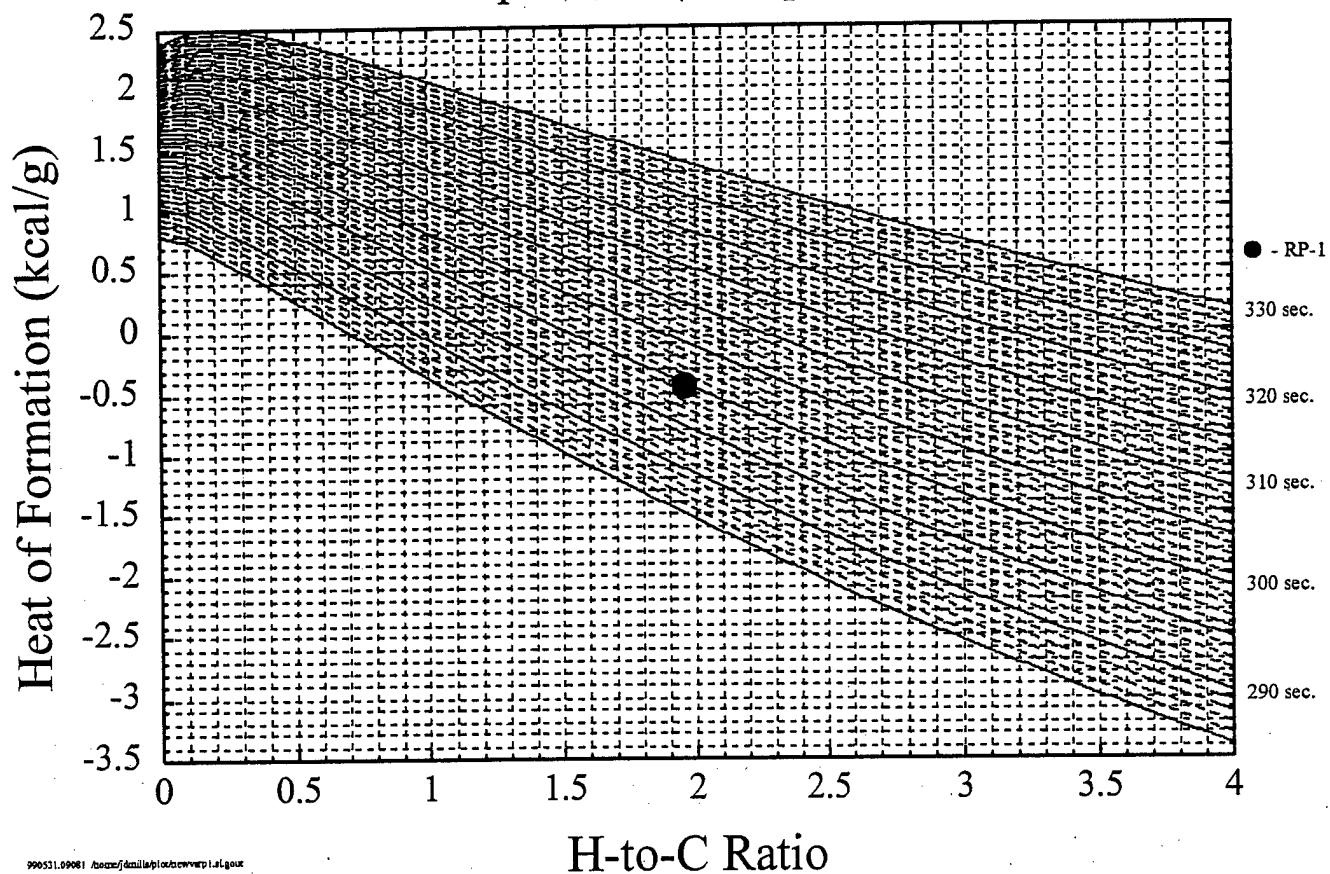
$\rho_f$ —fuel density

$a$ —density exponent (often, mission-specific constant)

Optimum Hydrocarbon  $I_{\{sp\}}$  vs. LOX  
Sea-Level Expansion (14.7 psi,  $P_c=1000$  psi)

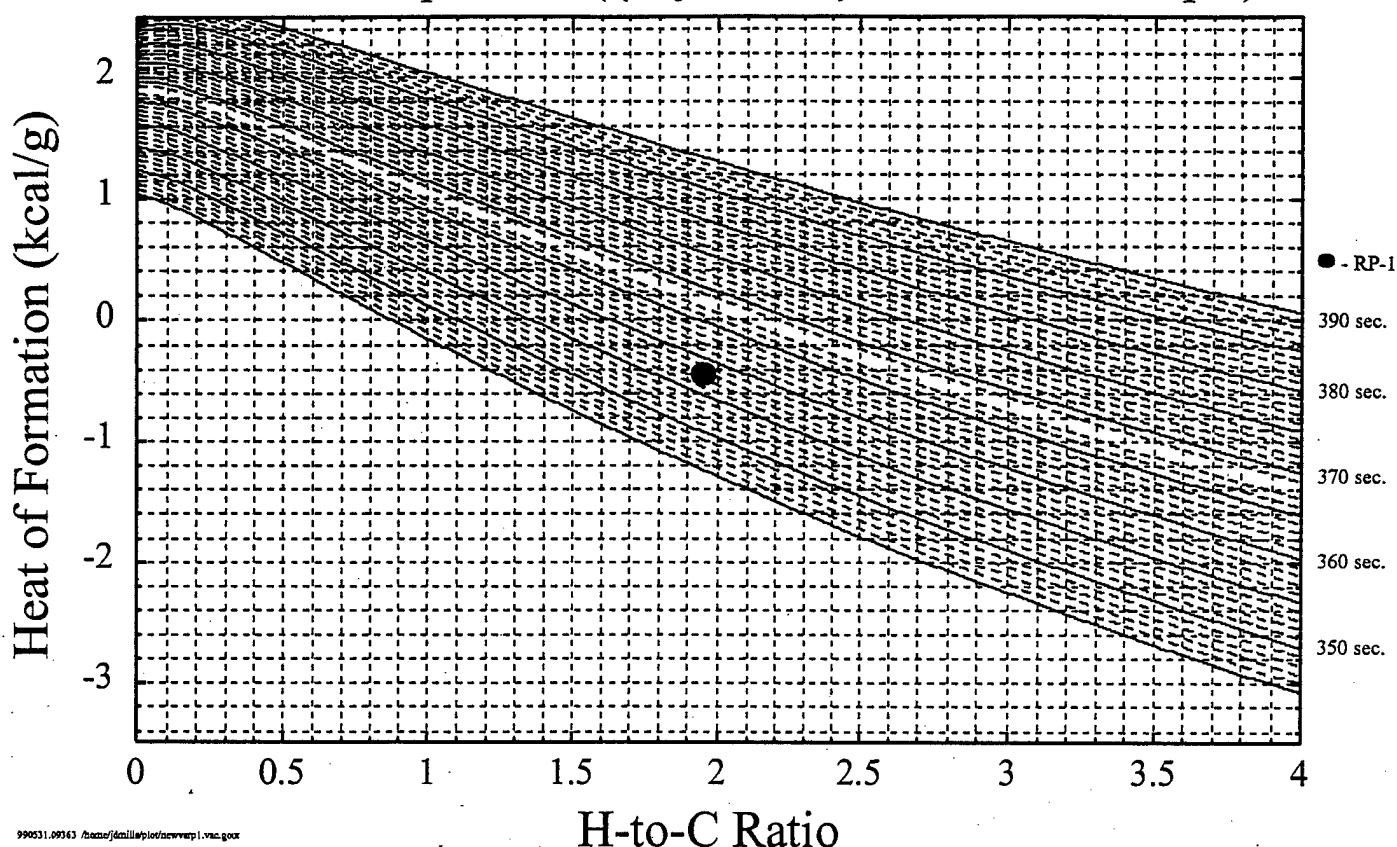


Optimum Hydrocarbon I\_{sp} vs. LOX  
Sea-Level Expansion (14.7 psi, P\_c=1000 psi)



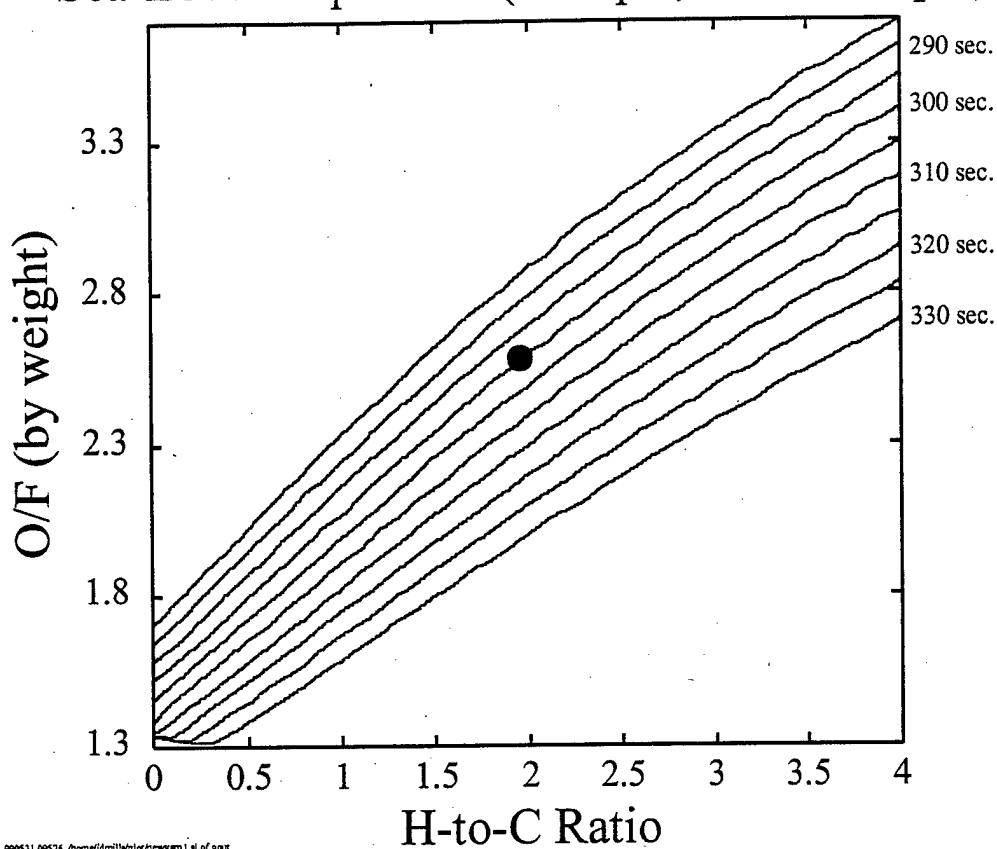
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Optimum Hydrocarbon L<sub>sp</sub> vs. LOX  
Vacuum Expansion ( $\{/\text{Symbol}\} = 40$ , P<sub>c</sub>=1000 psi)



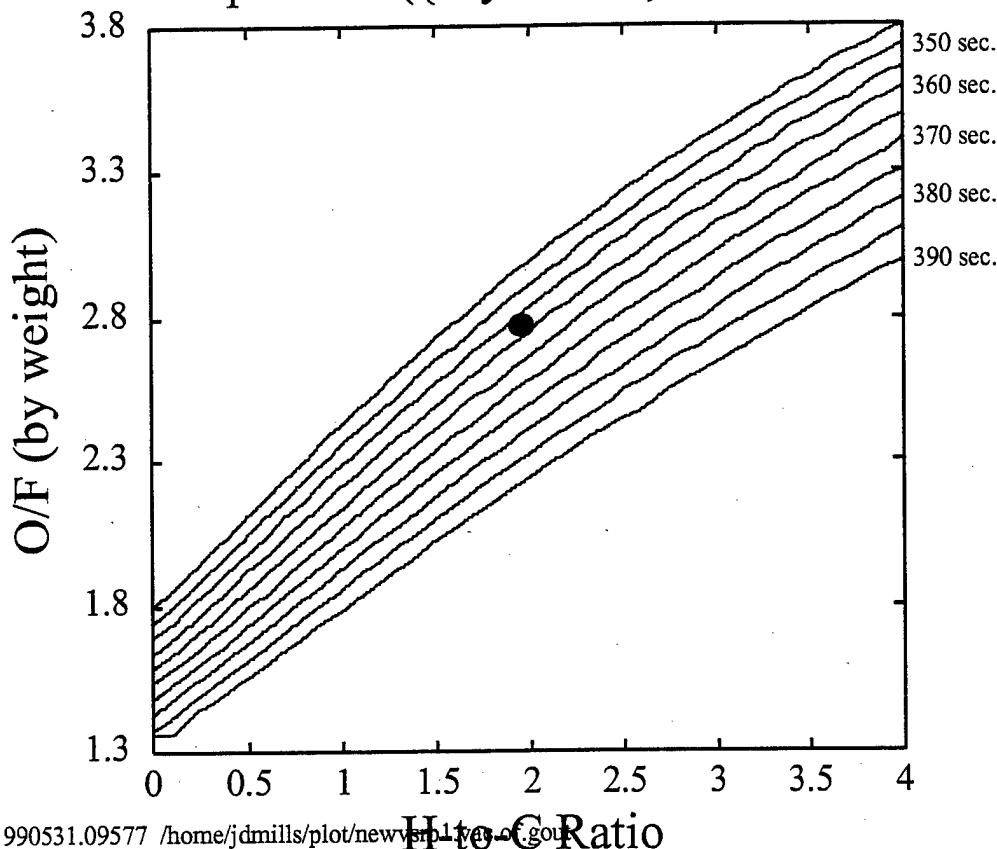
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O/F at Optimum I\_{sp}  
Sea-Level Expansion (14.7 psi, P\_c=1000 psi)



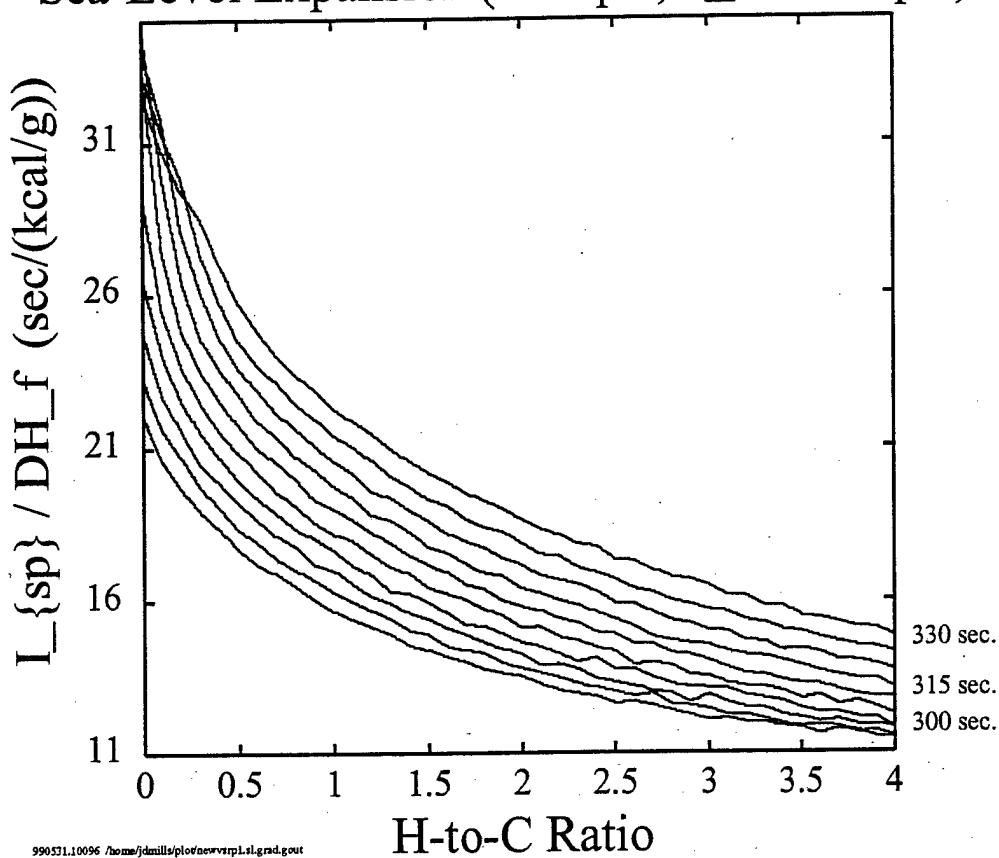
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O/F at Optimum I\_{sp}  
Vacuum Expansion ( $\{\text{Symbol e}\}=40$ ,  $P_c=1000 \text{ psi}$ )

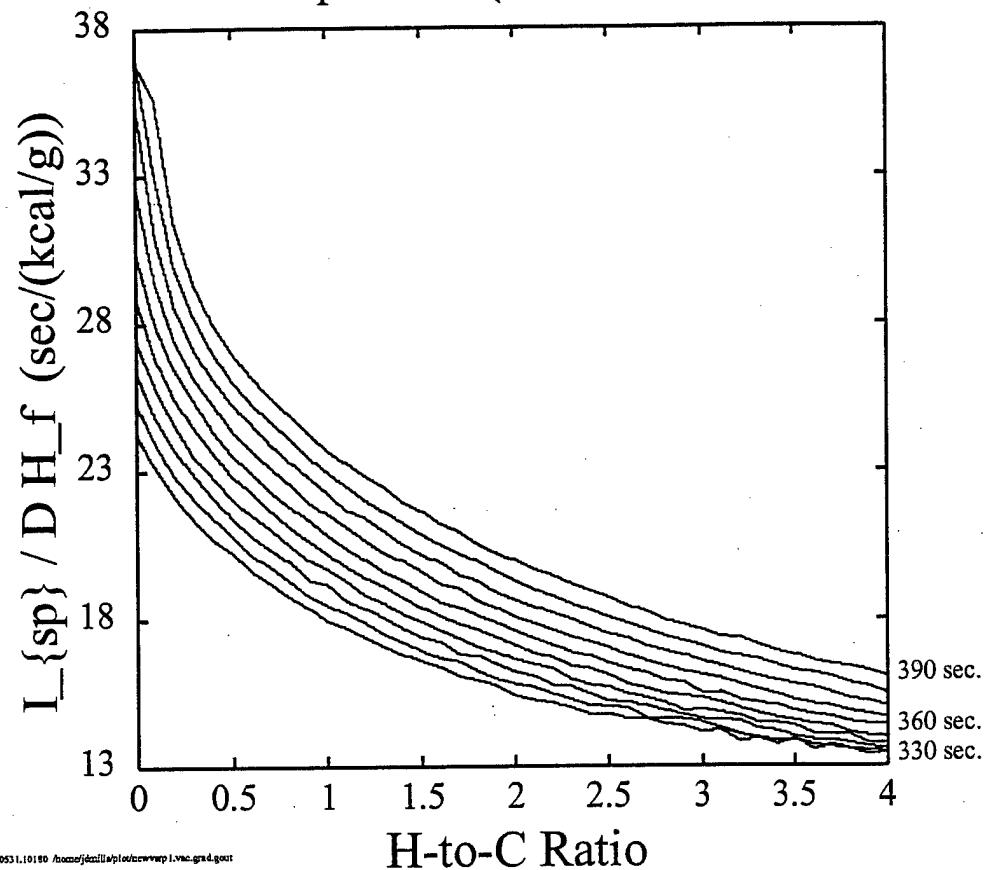


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Errors in Optimum  $I_{sp}$  with  $DH_f$   
Sea-Level Expansion (14.7 psi,  $P_c=1000$  psi)

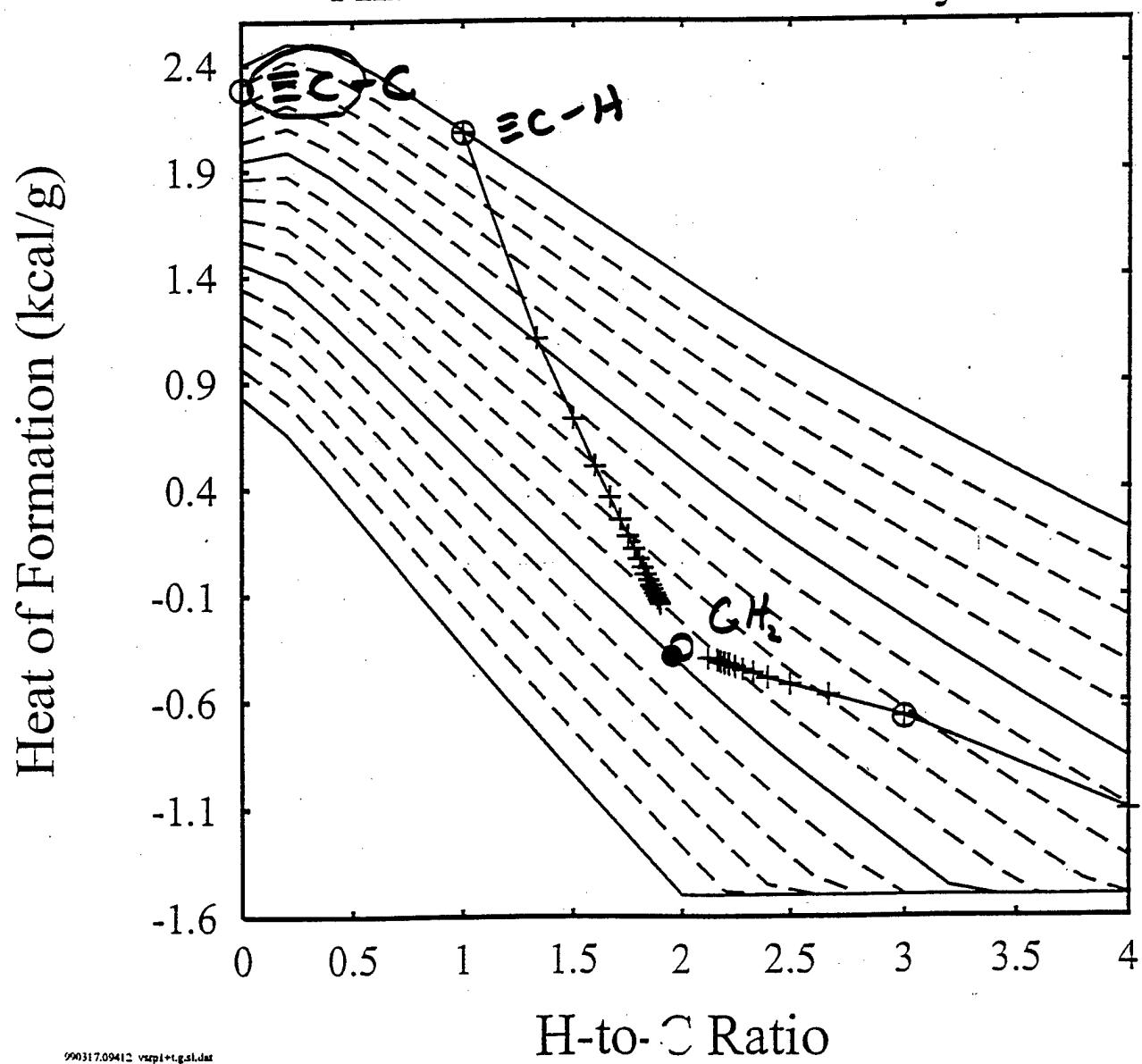


Errors in Optimum  $I_{\{sp\}}$  with  $DH_f$   
Vacuum Expansion ( $e=40$ ,  $P_c=1000$  psi)

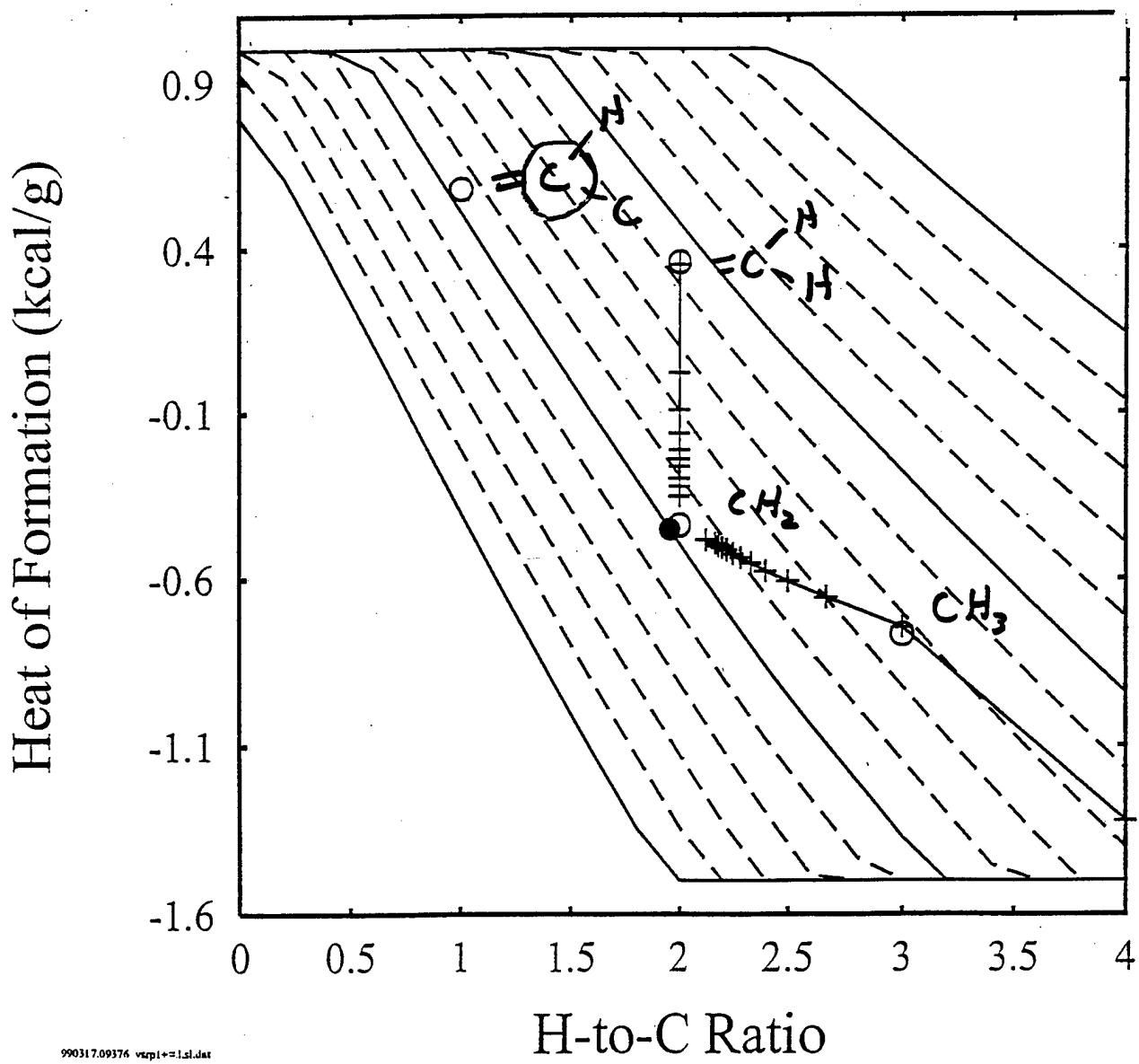


# A Specific-Impulse Survey of Chemical Families—

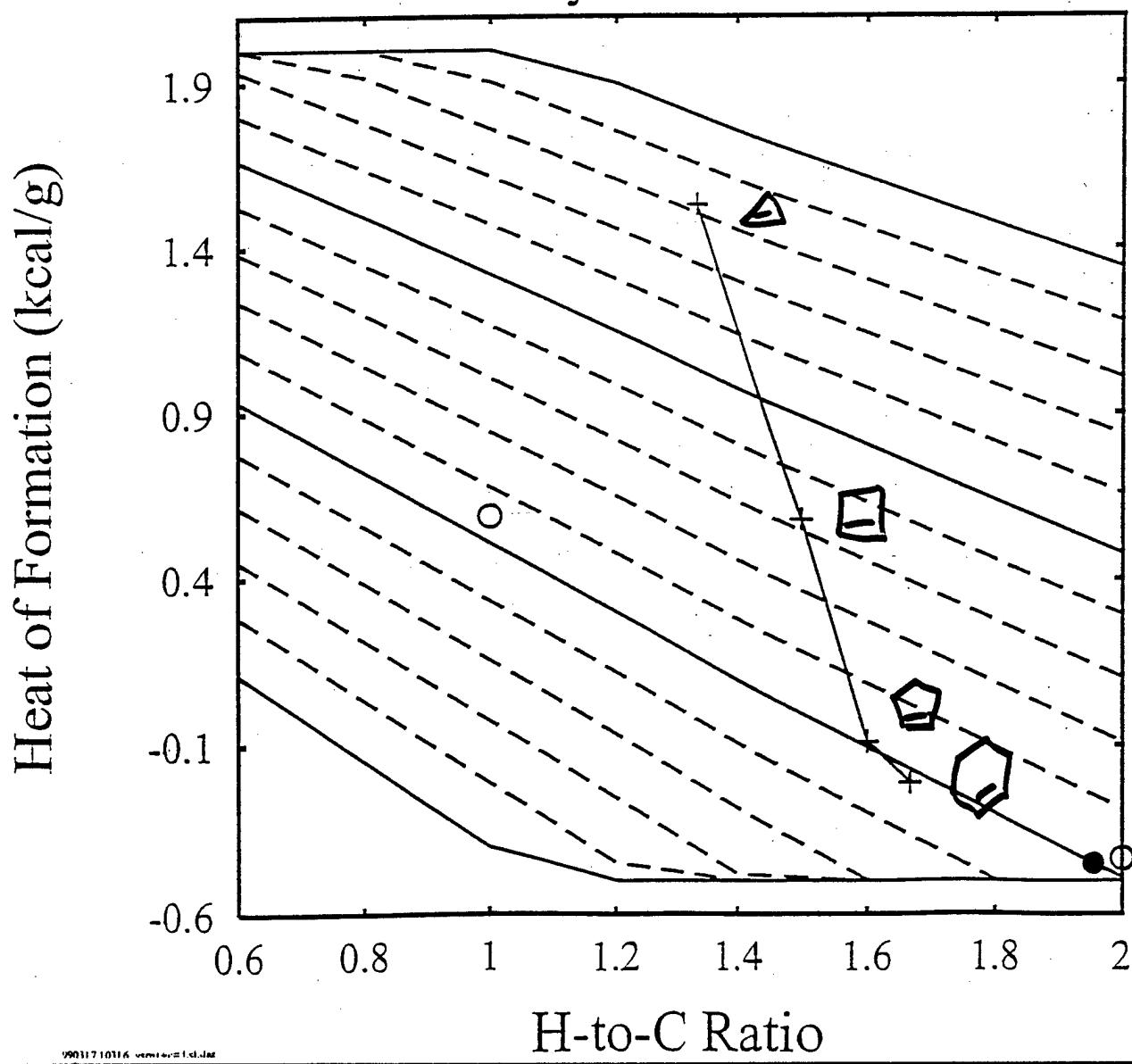
## Gas, Sea-Level Hydrocarbon Performance Alkanes and Terminal Alkynes



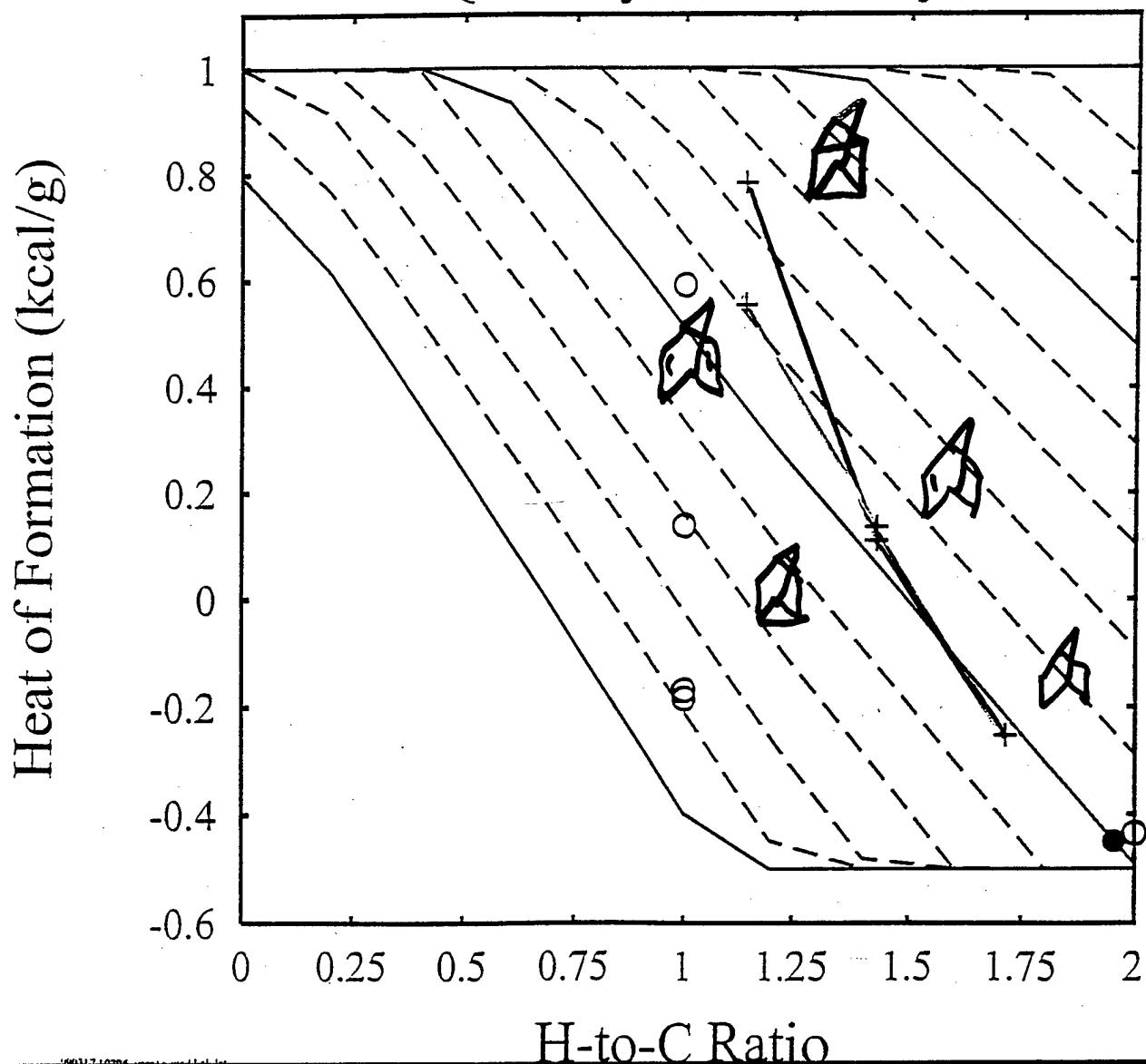
## Liquid, Sea-Level Hydrocarbon Performance Alkanes and Terminal Alkenes



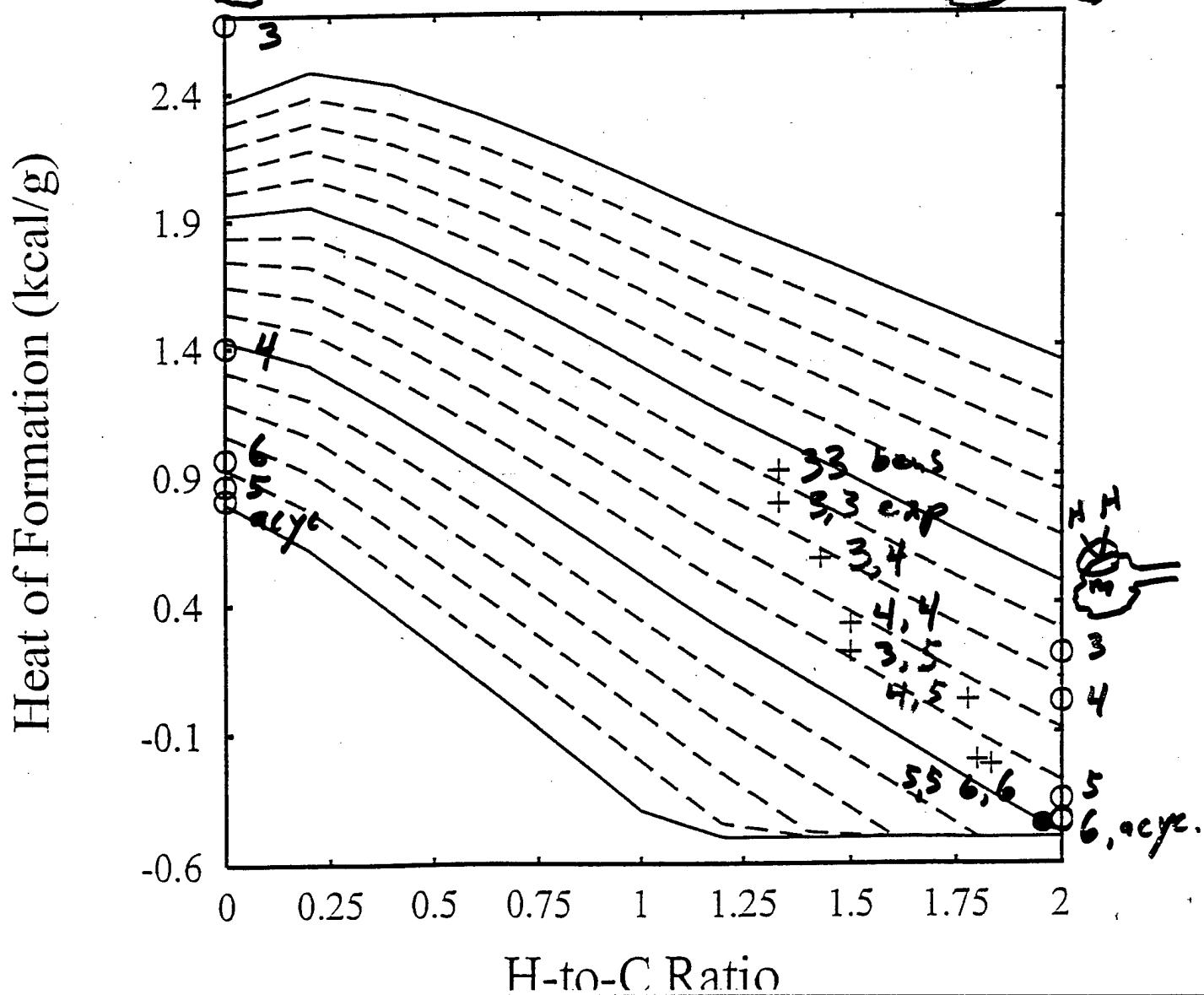
## Liquid, Sea-Level Hydrocarbon Performance Cycloalkenes



## Liquid, Sea-Level Hydrocarbon Performance Quadricyclane Family

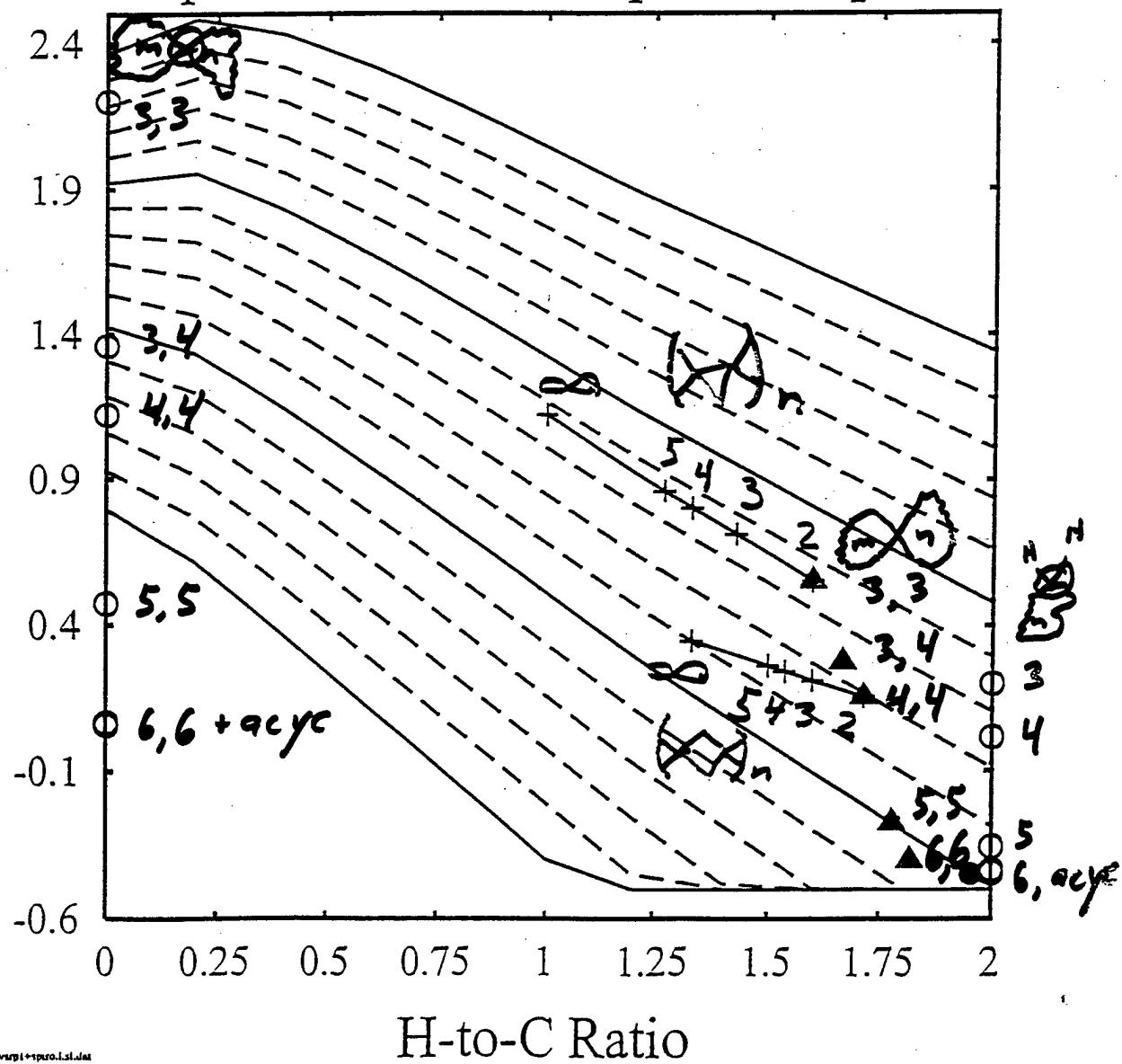


# Liquid, Sea-Level Hydrocarbon Performance Double-Bow Molecules

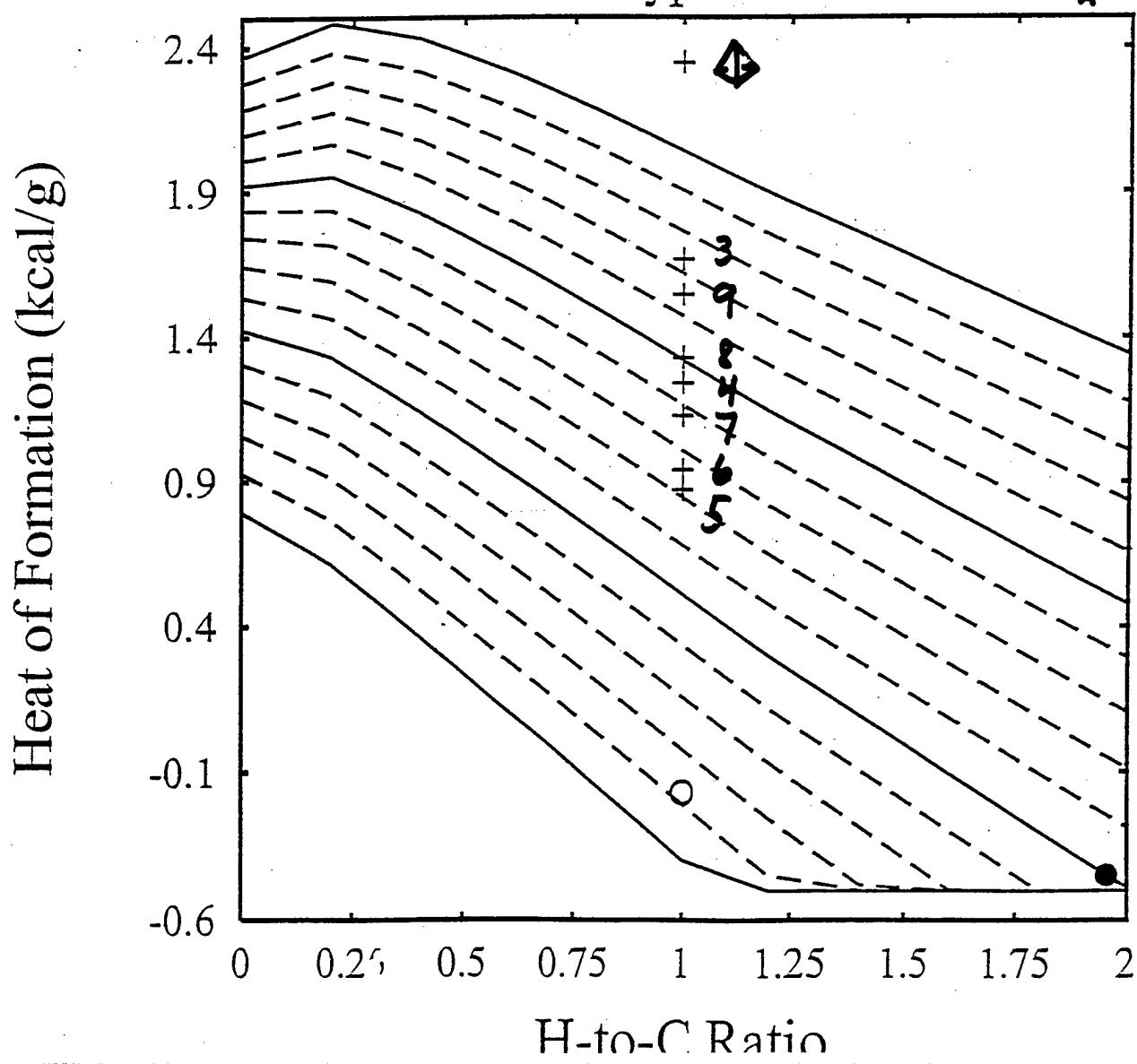


Heat of Formation (kcal/g)

## Liquid, Sea-Level Hydrocarbon Performance Simple and Catenated Spiro Compounds



Liquid, Sea-Level Hydrocarbon Performance  
Prismane-type Molecules



# “Benson-Group”, Specific Impulse—

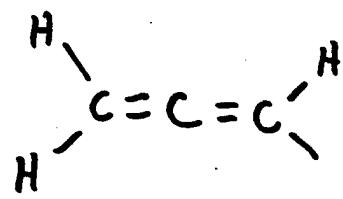
Additional Approximation: Benson-Group *Gedanken*  
Rocket

$$(I_{sp})_{opt} \approx \sum_i \frac{m_i}{m_{tot}} (I_{sp,i})_{opt}$$

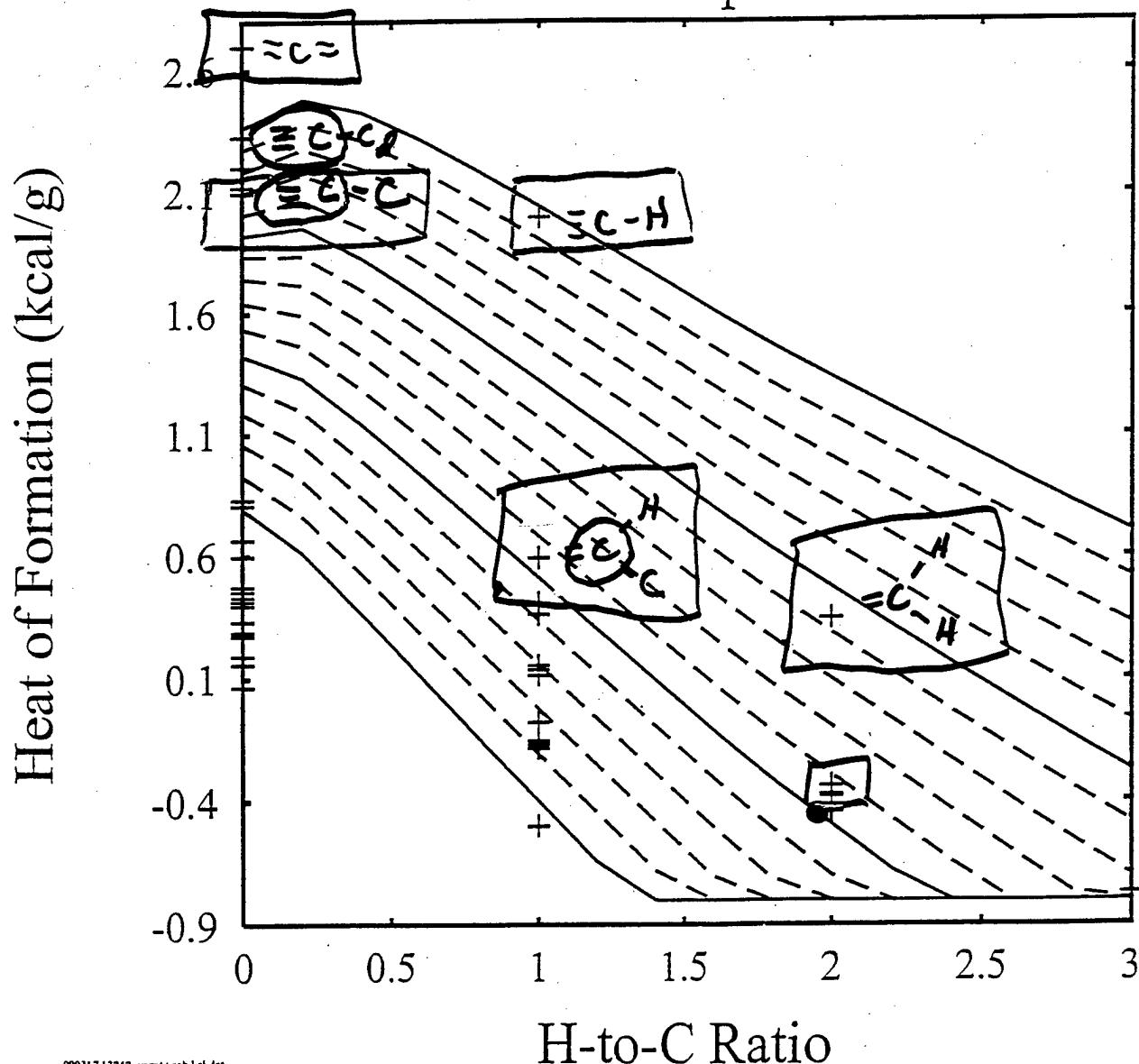
groups

$\frac{m_i}{m_{tot}}$ —group mass fraction

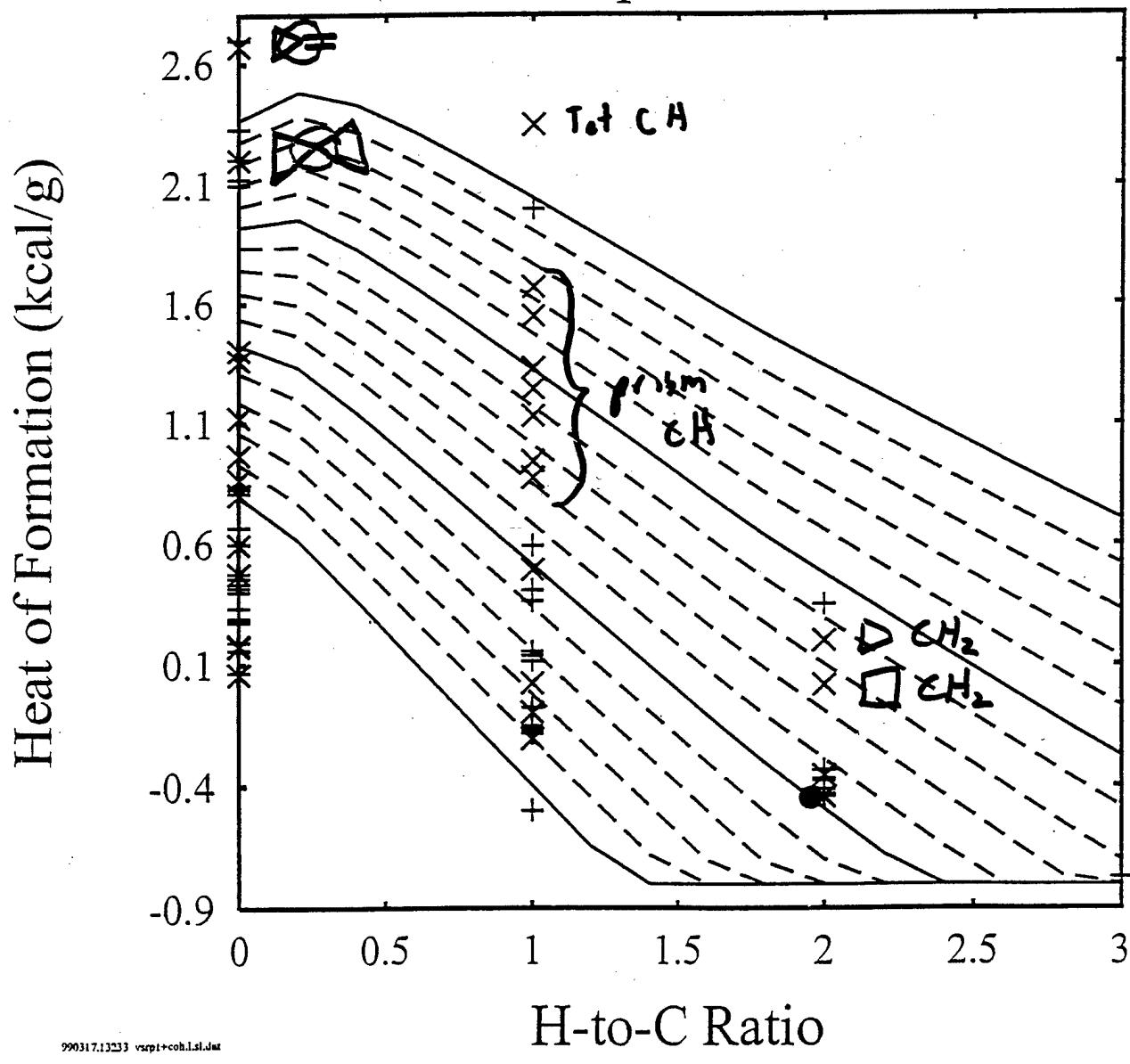
$(I_{sp,i})_{opt}$ —optimized specific impulse for the chemical group, alone



## Liquid, Sea-Level Hydrocarbon Performance Cohen Group Values

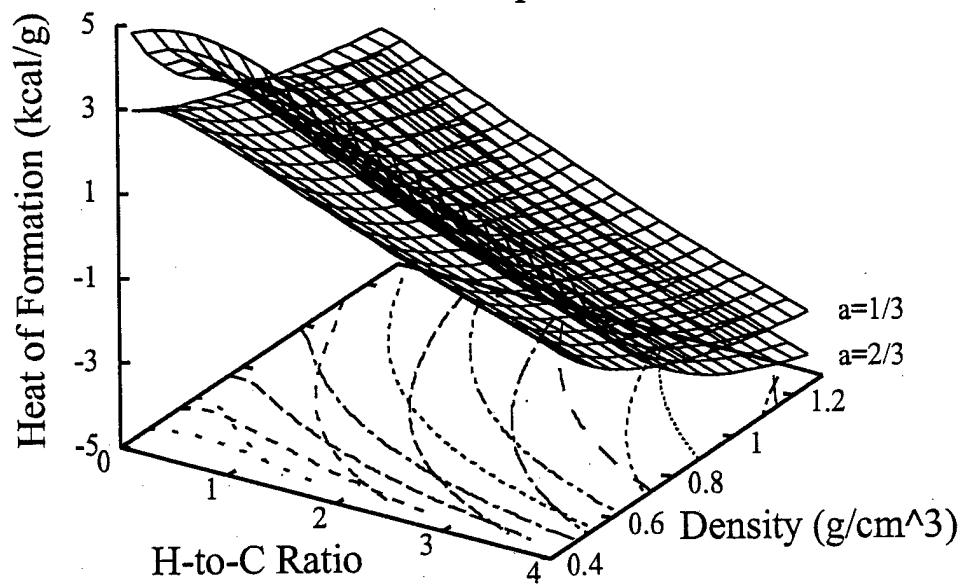


## Liquid, Sea-Level Hydrocarbon Performance Cohen Group Values + Extras



# **Effects of Fuel Density—**

Optimum Density  $I_{\{sp\}}$  vs. LOX  
Sea-Level Expansion



## Summary and Future Directions—

- Specific  $\Delta H_f$ ,  $r(\frac{H}{C})$ , and fuel density, alone, determine model, LOX-optimized rocket performance of hydrocarbon fuels
- Performance trade-offs among these parameters have been quantified and illustrated for ready reference
- Hydrocarbon fuels can be sought among known molecules in the parameterized performance space herein described
- The “Benson specific impulse” provides at least a semi-quantitative guide in the search for new, unknown rocket fuels

- Other chemical systems may be explored using these insights ( $C, H, N$ ; doped  $H_2$ )

## Acknowledgments—

These calculations were performed on local IBM RISC workstations using a modified version of the Air-Force specific-impulse rocket code written by C. Selph, R. Hall, C. Beckman, R. Acree, T. Magee and others.